## DEPARTMENT OF TRANSPORTATION FEDERAL AVIATION ADMINISTRATION

E-246 Revision 20 CONTINENTAL

E165-2, -3, -4 E185-1, -2, -3 (Military 0-470-7, -7A) E185-5, -8, -9 (Military 0-470-7B), -10, -11 March 1, 1974

## TYPE CERTIFICATE DATA SHEET NO. E-246

Engines of models described herein conforming with this data sheet (which is part of type certificate No. 246) and other approved data on file with the Federal Aviation Agency meet the minimum standards for use in certificated aircraft in accordance with pertinent aircraft data sheets and applicable portions of the Civil Air Regulations provided they are installed, operated and maintained as prescribed by the approved manufacturer's manuals and other approved instructions.

Type Certificate Holder

Teledyne Continental Motors P.O. Box 90 Mobile, Alabama 36601

Model	E165-2	E165-3	E165-4
Type	6НОА		
Rating			
Max. continuous hp.	165-2050		
r.p.m. at sea level pressure altitude			
Takeoff (5 min.) hp.	165-2050		
r.p.m. at full throttle			
Fuel (min. grade aviation gasoline)	80/87		
Lubricating oil,			
ambient air temperature, ° F.	Oil Grade		
Below 40	SAE 30		
Above 40	SAE 40		
Bore and Stroke, in.	5.00 x 4.00		
Displacement, cu. in.	471		
Compression ratio	7:1		
Weight (dry) lb.	351	352	344
Propeller Shaft			
SAE No.	4 (Flange)	20 (Spline)	
C. G. Location (basic engine)			
Fwd. of rear face, engine crankcase, in.	11.51		11.75
Below crankshaft centerline, in.	1.76		1.42
Beside crankshaft centerline toward	0.0		.11
1-3-5 side, in.			
Carburetion	Marvel-Schebler	Bendix-Stromberg	
	MA-4-5	PS-5C or PS-5CD	
Ignition, dual magnetos	2 Eisemann LA-6 or		
	Bendix-Scintilla S6LN-21		
	or 1 ea. S6LN-200 and -204		
Timing, ° BTC	Right 26, Left 26		
Spark Plugs	See NOTE 11.		
Oil sump capacity, quart	10 (7 usaable)		Dry
NOTES	1, 2, 3, 4, 6, 8, 10, 11		

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Model	E-185-1, -8, -11	E185-2	E185-3, -9, -10	E185-5
Туре	6НОА			
Rating				
Max. continuous hp.	185-2300			
r.p.m. at sea level pressure alt.				
Takeoff (5 min.) hp.	205-2600	185-2300	205-2600	185-2300
r.p.m. at full throttle				
Fuel (min. grade aviation gasoline)	80/87			
Lubricating oil,	Oil Grade			
ambient air temperature, ° F.				
Below 40	SAE 30			
Above 40	SAE 40			
Bore and Stroke, in.	5.00 x 4.00			
Displacement, cu. in.	471			
Compression ratio	7:1			
Weight (dry) lb.	344	351	352	343
Propeller Shaft				
SAE No.	20 (Spline)	4 (Flange)	20 (Spline)	4 (Flange)
C. G. Location (basic engine)			_	
Fwd. of rear face,	11.42	11.51	11.25	11.96
engine crankcase, in.				
Below crankshaft centerline, in.	1.75	1.76	1.06	1.39
Beside crankshaft centerline toward	.11	0.0	0.0	.11
1-3-5 side, in.				
Carburetion	Bendix-Stromberg	Marvel-Schebler	Bendix-Stromberg	
	PS-5C or PS-5CD	MA-4-5	PS-5C or PS-5CD	
Ignition, dual magnetos	2 Eisemann LA-6			
	or Bendix-Scintilla			
	S6LN-21 or 1227			
	or 1 ea. S6LN-200			
	and -204			
Timing, ° BTC	Right 26, Left 26			
Spark plugs				
Oil sump capacity, quart	Dry	10 (7 usable)		Dry
NOTES	1, 2, 3, 4, 5, 6, 7,	1, 2, 3, 4, 6, 8, 10,	1, 2, 3, 4, 5, 6, 7, 8,	1, 2, 3, 4, 5, 6,
	8, 10 , 11	11	9, 10, 11	8, 10, 11

<sup>&</sup>quot;- - -" Indicates "same as preceding model"

Certification Basis Part 13 of the Civil Air Regulations

Type Certificate No. 246

Production Basis Production Certificate No. 7. Production Certificate No. 508 (all models except

E185-10, 11).

NOTE 1. Maximum permissible cylinder head, barrel and oil inlet temperature, 525° F. (spark plug gasket), 315° F. and 225° F., respectively. Maximum permissible cylinder head temperature measured with well-type thermocouple is 450° F. (downdraft cooling only)

NOTE 2. Fuel inlet pressure limits are as follows:

Marvel-Schebler carburetor 1.5 to 9 p.s.i.g.

Stromberg-Carburetor 9 to 15 p.s.i.g.

Oil Pressure limits for normal operation:

1 ressure mints for normal operation

30 to 60 p.s.i.g.

NOTE 3. Accessories such as generators, starters, fuel and vacuum pumps, etc., previously listed in NOTE 3, are satisfactory for continued use with these engines. Such accessories are not integral engine accessories and, therefore, are not evaluated for approval during engine certification testing. The airworthiness of such accessories is substantiated during aircraft installation system approvals. The suitability of the engine mounting provisions as described below must be determined when processing such approvals.

Original	Type of	Direction	Speed Ratio	Max. Torque	(inlb.)	Maximum Overhang
Accessory	Drive Pad	of Rotation*	to Crankshaft	Continuous	Static	Moment (inlb.)
Generator	Automotive	CC	2.25:1	60**	800	100
Starter (dog type)	Aircraft	CC	1.15:1		2400	
(gear type)	Automotive	C	35.78:1			
Tachometer	AS 54	C	0.5:1			
Optional fuel pumps						
Right-hand	AND10000	C	1.364:1	90	600	65
Left-hand	AND10000	CC	1.667:1	25	400	25

<sup>\*</sup>C Clockwise. CC - Counterclockwise.

<sup>\*\*</sup> Generator torques listed eligible with two-piece rubber couplings CMC P/N 530998 or 626543. Early single-piece drive coupling P/N 25120 eligible for 7-½ in.-lb. maximum continuous and 50 in.-lb. maximum static torques. The Hartzell "T" Drive, P/N C-192, weighing 3 lb., is an approved dual-drive unit for installation on the above optional engine accessory drives with the following provisions:

Vacuum pump provision (optional)	
Maximum weight, lb.	4.5
Maximum overhang moment, inlb.	11.2
Maximum continuous torque, inlb.	16.0
Hydraulic pump provision (optional)	
Maximum weight, lb.	2.4
Maximum overhang moment, inlb.	7.0
Maximum continuous torque, inlb.	17.5

Hartzell Propeller Co. Installation Instruction No. 2 dated May 4, 1951, explains details for accomplishing installation of the C-192 "T" Drive.

- NOTE 4. Serial Nos. of engines equipped with dampered crankshafts are suffixed with the letter "D".
- NOTE 5. Engines with Serial Nos. below 5122 and those with undampered crankshafts are eligible for takeoff rating of only 185 b. hp. at 2300 r.p.m.
- NOTE 6. Fuel injection engines previously designated by the letter "J" have been deleted as they are no longer being manufactured. Fuel injectors used with these engines do not require a positive fuel head because they will draw fuel against an approximately 10-foot head.
- NOTE 7. E185-10 is not a production model but denotes E185-3 which has had Bendix Stromberg carburetor replaced with Marvel-Schebler MA-4-5. The E185-10 is not interchangeable with E185-3 installationwise, and is eligible for takeoff rating of only 185 b. hp. at 2300 r.p.m.

E-185-11 is identical to E185-8 except for revised mounting brackets.

E185-8 is identical to E185-1 and E185-9 is identical to E185-3 except for revised starter drive to accommodate dog rather than gear type starter.

Military 0-470-7, -7A engines are identical to E185-3. (0-470-7 incorporates provisions for 14 mm. spark plugs. When modified to incorporate 18 mm. spark plugs, it is designated as 0-470-7A). 0-470-B is identical to E185-9. These military engines are eligible for use with certificated aircraft and when so used, the engine nameplate should be revised to include the corresponding civil model designation and Type Certificate No.

NOTE 8. Carburetors of the same model may vary with the aircraft installation. The aircraft specification and /or engine manufacturer's manuals contain the carburetor parts list number (on carburetor nameplate) eligible for each aircraft installation.

NOTE 9. E185-9 with serial number suffixed by the letter "C" uses Marvel-Schebler MA-4-5 carburetor. These engines are not interchangeable, installationwise, with other E185-9 engines, and are eligible for takeoff rating of only 185 b. hp. at 2300 r.p.m.

NOTE 10. E-Series engines with the serial number suffixed by the letters "SP" utilize P/N 537733 cylinder. P/N 534895 piston and P/N 536938 and 536939 piston rings, and are interchangeable, otherwise, with standard engines of the same model.

NOTE 11. The following spark plugs are approved for use in these engines:

AC HSR83IR, SR83IR, HSR83P, SR83P, A88, HSR88, HS88, SR88, SR88D, S88D, S88

Auto Lite A4S, AH4S, SH2K, SH2M, H15, SH15, SH15R, SH20, SH20A, SH150, SH200A

BG RB485S, 613S, RB613S, 706, 706R, 706S, 706SR, 707SR, RB955S

Champion C26, RC26S, C27, C27S, REM38P, REM38W, RHM38P, RHM38W, RED39N,

REM39N, RHD39N, RHM39N, REM40E, RHM40E, D41N, ED41N, EM41N, M41N,

M42E, EM42E

Red Seal SA190, SE190, SJ190, SE230, SJ230

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